

# Computational Structural Mechanics And Dynamics



## Assignment – 1

Submitted by:

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# Assignment-1

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Computational Mechanics

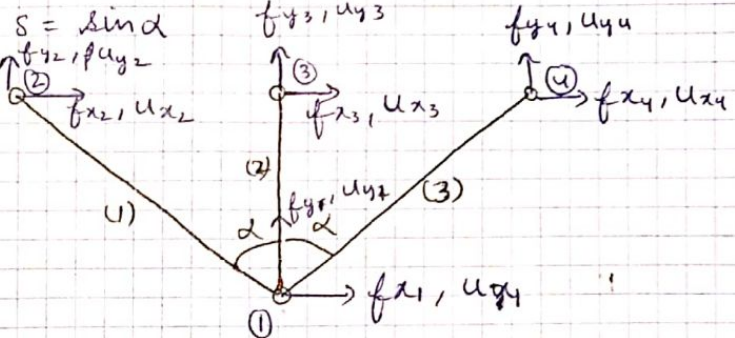
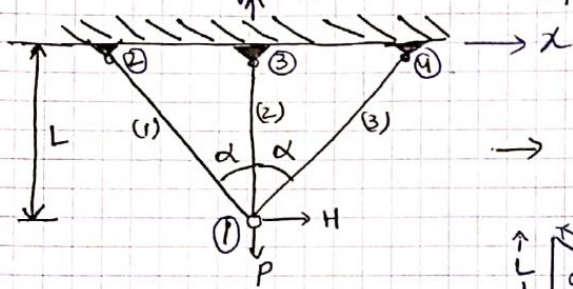
(a)  $\bar{u}^e = T^e u^e$  ;  $f^e = (T^e)^T \bar{f}^e$

$K^e = (T^e)^T K^e T^e$

$K^e = \frac{E^e A^e}{L^e} \begin{bmatrix} c^2 & sc & -c^2 & -sc \\ sc & s^2 & -sc & -s^2 \\ -c^2 & -sc & c^2 & sc \\ -sc & -s^2 & sc & s^2 \end{bmatrix}$

$C = \cos \alpha$

$S = \sin \alpha$



$\frac{L}{h} = \sin(90-\alpha) \Rightarrow h = \frac{L}{\cos \alpha} = \frac{L}{c}$

$L^{(1)} = L/c = L^{(3)}$  ;  $L^{(2)} = L$

$f^{(e)} = K^{(e)} u^{(e)} \rightarrow f_j$

Element (1)

$\begin{bmatrix} f_{x1}^{(1)} \\ f_{y1}^{(1)} \\ f_{x2}^{(1)} \\ f_{y2}^{(1)} \end{bmatrix} = \frac{EA}{L/c} \begin{bmatrix} s^2 & -sc & -s^2 & sc \\ -sc & c^2 & sc & -c^2 \\ -s^2 & sc & s^2 & -sc \\ sc & -c^2 & -sc & c^2 \end{bmatrix} \begin{bmatrix} u_{x1}^{(1)} \\ u_{y1}^{(1)} \\ u_{x2}^{(1)} \\ u_{y2}^{(1)} \end{bmatrix}$

Element (2)

$K^{(1)} = \frac{EA}{L} \begin{bmatrix} cs^2 & -sc^2 & -cs^2 & sc^2 \\ -sc^2 & c^3 & sc^2 & -c^3 \\ -cs^2 & sc^2 & cs^2 & -sc^2 \\ sc^2 & -c^3 & -sc^2 & c^3 \end{bmatrix} \begin{matrix} | 1 \\ | 2 \end{matrix}$

Element (2)

$K^{(2)} = \frac{EA}{L} \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & -1 \\ 0 & 0 & 0 & 0 \\ 0 & -1 & 0 & 1 \end{bmatrix} \begin{matrix} | 1 \\ | 3 \end{matrix}$

Element (3)

$K^{(3)} = \frac{EA}{L/c} \begin{bmatrix} s^2 & sc & -s^2 & sc \\ sc & c^2 & -sc & -c^2 \\ -s^2 & -sc & s^2 & sc \\ -sc & -c^2 & sc & c^2 \end{bmatrix} \begin{matrix} | 1 \\ | 4 \end{matrix}$   
 $= \frac{EA}{L} \begin{bmatrix} cs^2 & sc^2 & -cs^2 & -sc^2 \\ sc^2 & c^3 & -sc^2 & -c^3 \\ -cs^2 & -sc^2 & cs^2 & sc^2 \\ -sc^2 & -c^3 & sc^2 & c^3 \end{bmatrix} \begin{matrix} | 1 \\ | 4 \end{matrix}$

$K = K^{(1)} + K^{(2)} + K^{(3)} = \begin{bmatrix} K_{11}^{(1)} + K_{11}^{(2)} + K_{11}^{(3)} & K_{12}^{(1)} & K_{13}^{(2)} & K_{14}^{(3)} \\ K_{21}^{(1)} & K_{22}^{(1)} & 0 & 0 \\ K_{31}^{(2)} & 0 & K_{33}^{(2)} & 0 \\ K_{41}^{(3)} & 0 & 0 & K_{44}^{(3)} \end{bmatrix}$



$$K = \frac{EA}{L} \begin{bmatrix} 2cs^2 & 0 & -cs^2 & sc^2 & 0 & 0 & -cs^2 & -sc^2 \\ 0 & 2c^3+1 & sc^2 & -c^3 & 0 & -1 & -sc^2 & -c^3 \\ -cs^2 & sc^2 & cs^2 & -sc^2 & 0 & 0 & 0 & 0 \\ sc^2 & -c^3 & -sc^2 & c^3 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & -1 & 0 & 0 & 0 & 1 & 0 & 0 \\ -cs^2 & -sc^2 & 0 & 0 & 0 & 0 & cs^2 & sc^2 \\ -sc^2 & -c^3 & 0 & 0 & 0 & 0 & sc^2 & c^3 \end{bmatrix}$$

$$K = \frac{EA}{L} \begin{bmatrix} 2cs^2 & 0 & -cs^2 & c^2s & 0 & 0 & -cs^2 & -c^2s \\ & 1+2c^3 & c^2s & -c^3 & 0 & -1 & -c^2s & -c^3 \\ & & cs^2 & -c^2s & 0 & 0 & 0 & 0 \\ & & & c^3 & 0 & 0 & 0 & 0 \\ & & & & 0 & 0 & 0 & 0 \\ & & & & & 1 & 0 & 0 \\ & & & & & & 0 & 0 \\ & & & & & & cs^2 & c^2s \\ & & & & & & c^2s & c^3 \end{bmatrix}$$

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From the above matrix, it can be seen that 5<sup>th</sup> row and column contain only 0 because 5<sup>th</sup> row and column represents x-direction component of 3<sup>rd</sup> mode and bar (2) is vertical. ~~Therefore~~, Bar (2) is loaded axially and axial direction in this case is along y-axis. So, Row 5<sup>th</sup> and Column 5<sup>th</sup> are zeroes.

$$f = K u$$

$$\begin{bmatrix} H \\ -P \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix} = \frac{EA}{L} \begin{bmatrix} 2cs^2 & 0 & -cs^2 & c^2s & 0 & 0 & -cs^2 & -c^2s \\ & 1+2c^3 & c^2s & -c^3 & 0 & -1 & -c^2s & -c^3 \\ & & cs^2 & -c^2s & 0 & 0 & 0 & 0 \\ & & & c^3 & 0 & 0 & 0 & 0 \\ & & & & 0 & 0 & 0 & 0 \\ & & & & & 1 & 0 & 0 \\ & & & & & & 0 & 0 \\ & & & & & & cs^2 & c^2s \\ & & & & & & c^2s & c^3 \end{bmatrix} \begin{bmatrix} u_{x1} \\ u_{y1} \\ u_{x2} \\ u_{y2} \\ u_{x3} \\ u_{y3} \\ u_{x4} \\ u_{y4} \end{bmatrix}$$



(b) 2, 3 and 4 Nodes are fixed  $\therefore$  displacement will be 0 at these nodes.

$$u_{x2} = u_{y2} = u_{x3} = u_{y3} = u_{x4} = u_{y4} = 0$$

$\Rightarrow$  There will be no contribution to global force vector due to displacement BCs.

$\therefore$  Modified Stiffness matrix is.

$$\frac{EA}{L} \begin{bmatrix} 2cs^2 & 0 \\ 0 & 1+2c^3 \end{bmatrix} \begin{bmatrix} u_{x1} \\ u_{y1} \end{bmatrix} = \begin{bmatrix} H \\ -P \end{bmatrix}$$

$$(c) \quad \frac{EA}{L} (2cs^2) u_{x1} = H \quad \Rightarrow \quad u_{x1} = \left( \frac{L}{EA} \right) \left( \frac{H}{2cs^2} \right)$$

$$\frac{EA}{L} (1+2c^3) u_{y1} = -P \quad \Rightarrow \quad u_{y1} = \left( \frac{L}{EA} \right) \left( \frac{-P}{1+2c^3} \right)$$

$$\alpha \rightarrow 0 \Rightarrow c \rightarrow 1 \quad \text{and} \quad s \rightarrow 0 \quad \Rightarrow \quad cs^2 \rightarrow 0$$

$$\therefore u_{x1} \rightarrow \infty \quad \text{and} \quad u_{y1} \rightarrow \frac{-PL}{3EA}$$

When  $\alpha \rightarrow 0$ , nodes 2, 3 and 4 tends to overlap, so instead of 3 bar we can consider a single bar with very high stiffness indicating high resistance to motion due to external force. But, there will be no resistance to horizontal force  $H$  ( $x$ -axis) as bars (1) and (3) are now vertical; therefore  $u_{x1}$  blows up.

$$\alpha \rightarrow \pi/2 \Rightarrow c \rightarrow 0 \quad \text{and} \quad s \rightarrow 1 \Rightarrow cs^2 \rightarrow 0$$

$$\therefore u_{x1} \rightarrow \infty \quad \text{and} \quad u_{y1} = \left( \frac{L}{EA} \right) (-P)$$

When  $\alpha \rightarrow \frac{\pi}{2}$ , length of bar (1) and bar (3) will tend to infinity. Therefore, Nodes 1 and 3 will <sup>tend to</sup> be coincident.

So,  $u_{x1}$  will ~~be~~ tend to infinity ~~as~~ ~~stiffness~~ ~~tends~~ ~~to~~ ~~0~~ and makes no physical sense.



$$(d) \bar{u}^e = T^e u^e \quad T^e = \begin{bmatrix} c & s & 0 & 0 \\ -s & c & 0 & 0 \\ 0 & 0 & c & s \\ 0 & 0 & -s & c \end{bmatrix}$$

Element (1)  $\theta = \alpha + 90^\circ$

$$\bar{u}^{(1)} = T^{(1)} u^{(1)} \Rightarrow \begin{bmatrix} \bar{u}_{x1} \\ \bar{u}_{y1} \\ \bar{u}_{x2} \\ \bar{u}_{y2} \end{bmatrix} = \begin{bmatrix} -s & c & 0 & 0 \\ -c & -s & 0 & 0 \\ 0 & 0 & -s & c \\ 0 & 0 & -c & -s \end{bmatrix} \begin{bmatrix} u_{x1} \\ u_{y1} \\ 0 \\ 0 \end{bmatrix} = \begin{bmatrix} -s u_{x1} + c u_{y1} \\ -c u_{x1} - s u_{y1} \\ 0 \\ 0 \end{bmatrix}$$

$$d^{(1)} = \bar{u}_{x2} - \bar{u}_{x1} = 0 - (-s u_{x1} + c u_{y1}) = s u_{x1} - c u_{y1}$$

$$= s \left( \frac{L}{EA} \right) \left( \frac{H}{2cs^2} \right) - c \left( \frac{L}{EA} \right) \left( \frac{-P}{1+2c^3} \right)$$

$$= \left( \frac{L}{EA} \right) \left[ \frac{H}{2cs} + \frac{Pc}{1+2c^3} \right]$$

$$F^{(1)} = \frac{EA}{L} c \times \frac{L}{EA} \left[ \frac{H}{2cs} + \frac{Pc}{1+2c^3} \right] = \frac{H}{2s} + \frac{Pc^2}{1+2c^3}$$

Element (2)  $\theta = 90^\circ$

$$\bar{u}^{(2)} = T^{(2)} u^{(2)} \Rightarrow \begin{bmatrix} \bar{u}_{x1} \\ \bar{u}_{y1} \\ \bar{u}_{x3} \\ \bar{u}_{y3} \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ -1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & -1 \end{bmatrix} \begin{bmatrix} u_{x1} \\ u_{y1} \\ u_{x3} \\ u_{y3} \end{bmatrix} = \begin{bmatrix} u_{y1} \\ -u_{x1} \\ 0 \\ 0 \end{bmatrix}$$

$$d^{(2)} = \bar{u}_{x3} - \bar{u}_{x1} = 0 - u_{y1} = \left( \frac{L}{EA} \right) \left( \frac{-P}{1+2c^3} \right)$$

$$F^{(2)} = \frac{EA}{L} \left( \frac{L}{EA} \right) \left( \frac{P}{1+2c^3} \right) = \frac{P}{1+2c^3}$$

Element (3)  $\theta = 90 - \alpha$

$$\bar{u}^{(3)} = T^{(3)} u^{(3)} \Rightarrow \begin{bmatrix} \bar{u}_{x1} \\ \bar{u}_{y1} \\ \bar{u}_{x4} \\ \bar{u}_{y4} \end{bmatrix} = \begin{bmatrix} s & c & 0 & 0 \\ -c & s & 0 & 0 \\ 0 & 0 & s & c \\ 0 & 0 & -c & s \end{bmatrix} \begin{bmatrix} u_{x1} \\ u_{y1} \\ 0 \\ 0 \end{bmatrix} = \begin{bmatrix} s u_{x1} + c u_{y1} \\ -c u_{x1} + s u_{y1} \\ 0 \\ 0 \end{bmatrix}$$

$$d^{(3)} = \bar{u}_{x4} - \bar{u}_{x1} = 0 - (s u_{x1} + c u_{y1})$$

$$= -s \left( \frac{L}{EA} \right) \left( \frac{H}{2cs^2} \right) + c \left( \frac{L}{EA} \right) \left( \frac{P}{1+2c^3} \right)$$

$$= \frac{L}{EA} \left[ \frac{-H}{2cs} + \frac{Pc}{1+2c^3} \right]$$

$$F^{(3)} = \frac{EA}{L} c \times \frac{L}{EA} \left[ \frac{-H}{2cs} + \frac{Pc}{1+2c^3} \right] = \frac{-H}{2s} + \frac{Pc^2}{1+2c^3}$$

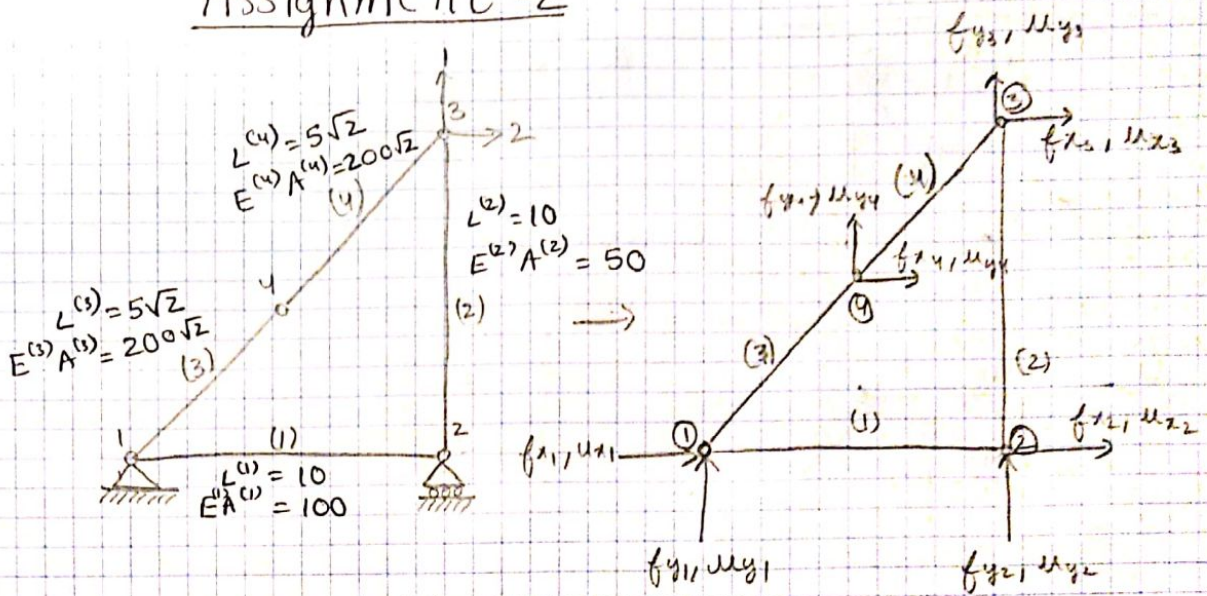
If  $\alpha \rightarrow 0$ ,  $s \rightarrow 0$  therefore  $F^{(1)} \rightarrow \infty$  and  $F^{(3)} \rightarrow \infty$   
and  $H \neq 0$

when  $\alpha \rightarrow 0$ , bars 1, 2 and 3 are coincident, i.e., a vertical truss member. Trusses <sup>members</sup> can only allow axial loading.

when a horizontal force  $H$  is applied, the truss member does take it into account and hence  $F^{(1)}$  and  $F^{(3)}$  blows up



# Assignment - 2



$$K^{(e)} = \begin{bmatrix} c^2 & sc & -c^2 & -sc \\ sc & s^2 & -sc & -s^2 \\ -c^2 & -sc & c^2 & sc \\ -sc & -s^2 & sc & s^2 \end{bmatrix}$$

$$K^{(1)} = 10 \begin{bmatrix} 1 & 0 & -1 & 0 \\ 0 & 0 & 0 & 0 \\ -1 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$$

$$K^{(2)} = 5 \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & -1 \\ 0 & 0 & 0 & 0 \\ 0 & -1 & 0 & 1 \end{bmatrix}$$

$$K^{(3)} = 40 \begin{bmatrix} 0.5 & 0.5 & -0.5 & -0.5 \\ 0.5 & 0.5 & -0.5 & -0.5 \\ -0.5 & -0.5 & 0.5 & 0.5 \\ -0.5 & -0.5 & 0.5 & 0.5 \end{bmatrix} = 20 \begin{bmatrix} 1 & 1 & -1 & -1 \\ 1 & 1 & -1 & -1 \\ -1 & -1 & 1 & 1 \\ -1 & -1 & 1 & 1 \end{bmatrix}$$

$$K^{(4)} = 20 \begin{bmatrix} 1 & 1 & -1 & -1 \\ 1 & 1 & -1 & -1 \\ -1 & -1 & 1 & 1 \\ -1 & -1 & 1 & 1 \end{bmatrix}$$

$$\text{Global } K = \begin{bmatrix} 30 & 20 & -10 & 0 & 0 & 0 & -20 & -20 \\ 20 & 20 & 0 & 0 & 0 & 0 & -20 & -20 \\ -10 & 0 & 10 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0.5 & 0 & -5 & 0 & 0 \\ 0 & 0 & 0 & 0 & 20 & 20 & -20 & -20 \\ 0 & 0 & 0 & -5 & 20 & 25 & -20 & -20 \\ -20 & 20 & 0 & 0 & -20 & -20 & 40 & 40 \\ -20 & 20 & 0 & 0 & -20 & -20 & 40 & 40 \end{bmatrix}$$

Now  $u_{x1} = u_{y1} = 0$        $u_{y2} = 0$   
 $f_{x3} = 2$        $f_{y3} = 1$



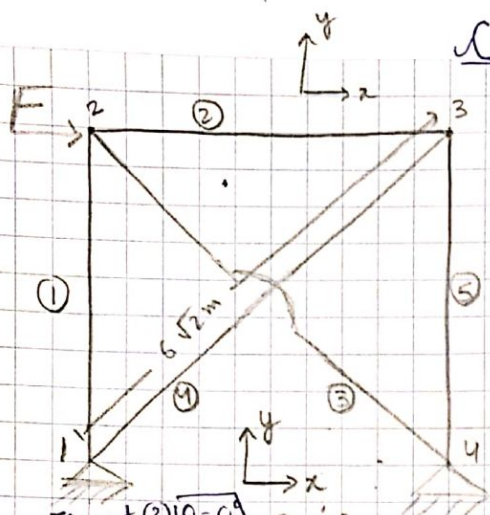
$K u = f$   
 Applying these B.C., we get

$$\begin{bmatrix} 10 & 0 & 0 & 0 & 0 \\ 0 & 20 & 20 & -20 & -20 \\ 0 & 20 & 25 & -20 & -20 \\ 0 & -20 & -20 & 40 & 40 \\ 0 & -20 & -20 & 40 & 40 \end{bmatrix}
 \begin{bmatrix} u_{x2} \\ u_{y3} \\ u_{y3} \\ u_{x4} \\ u_{y4} \end{bmatrix}
 =
 \begin{bmatrix} 0 \\ 2 \\ 1 \\ 0 \\ 0 \end{bmatrix}$$

- This Master stiffness matrix is singular as its determinant is zero.
- Physically, truss members ~~can't~~ take ~~shear~~ loads ~~and~~ bending moment. and used in a ~~truss~~ completely triangulated trusses. So, matrix is singular and this ~~is~~ 4-noded truss is unstable.



# classwork



$L = 6m$  ,  $E = 200 \text{ GPa} = 20 \times 10^9 \text{ Pa}$

$F = 80 \text{ kN} = 8 \times 10^4 \text{ N}$  ,  $A = 6 \text{ cm} = 6 \times 10^{-2} \text{ m}$

Element (1)  $\theta = 90^\circ$

$$k^{(1)} = \frac{2 \times 10^9 \times 6 \times 10^{-2}}{6} \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & -1 \\ 0 & 0 & 0 & 0 \\ 0 & -1 & 0 & 1 \end{bmatrix} = 2 \times 10^7 \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & -1 \\ 0 & 0 & 0 & 0 \\ 0 & -1 & 0 & 1 \end{bmatrix}$$

Element (2)  $\theta = 0^\circ$

$$k^{(2)} = 2 \times 10^7 \begin{bmatrix} 1 & 0 & -1 & 0 \\ 0 & 0 & 0 & 0 \\ -1 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$$

Element (5)  $\theta = 90^\circ$

$$k^{(5)} = 2 \times 10^7 \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & -1 \\ 0 & 0 & 0 & 0 \\ 0 & -1 & 0 & 1 \end{bmatrix}$$

Element 3  $\theta = 135^\circ$

$$k^{(3)} = \frac{2 \times 10^7}{\sqrt{2}} \times \frac{1}{2} \begin{bmatrix} 1 & -1 & -1 & 1 \\ -1 & 1 & 1 & -1 \\ -1 & 1 & 1 & -1 \\ 1 & -1 & -1 & 1 \end{bmatrix} = \frac{10^7}{\sqrt{2}} \begin{bmatrix} 1 & -1 & -1 & 1 \\ -1 & 1 & 1 & -1 \\ -1 & 1 & 1 & -1 \\ 1 & -1 & -1 & 1 \end{bmatrix}$$

Element 4  $\theta = 45^\circ$

$$k^{(4)} = \frac{2 \times 10^7}{\sqrt{2}} \times \frac{1}{2} \begin{bmatrix} 1 & 1 & -1 & -1 \\ 1 & 1 & -1 & -1 \\ -1 & -1 & 1 & 1 \\ -1 & -1 & 1 & 1 \end{bmatrix} = \frac{10^7}{\sqrt{2}} \begin{bmatrix} 1 & 1 & -1 & -1 \\ 1 & 1 & -1 & -1 \\ -1 & -1 & 1 & 1 \\ -1 & -1 & 1 & 1 \end{bmatrix}$$

Global Matrix  $K = 10^7 \times \begin{bmatrix} 0.707 & 0.707 & 0 & 0 & -0.707 & -0.707 & 0 & 0 \\ 0.707 & 2.707 & 0 & -2 & -0.707 & -0.707 & 0 & 0 \\ 0 & 0 & 2.707 & -0.707 & -2 & 0 & -0.707 & 0.707 \\ 0 & -2 & -0.707 & 2.707 & 0 & 0 & 0.707 & -0.707 \\ -0.707 & -0.707 & -2 & 0 & 2.707 & 0.707 & 0 & 0 \\ -0.707 & -0.707 & 0 & 0 & 0.707 & 2.707 & 0 & -2 \\ 0 & 0 & -0.707 & 0.707 & 0 & 0 & 0.707 & -0.707 \\ 0 & 0 & 0.707 & -0.707 & 0 & -2 & -0.707 & 2.707 \end{bmatrix}$

## B.C.s

$u_{x1} = u_{y1} = u_{x4} = u_{y4} = 0$  ;  $f_{x2} = 8 \times 10^4 \text{ N}$  ;  $f_{y2} = f_{x3} = f_{y3} = 0$

Imposing these B.C.s, we get  $f = Ku$

$$\begin{bmatrix} 8 \times 10^4 \\ 0 \\ 0 \\ 0 \end{bmatrix} = 10^7 \begin{bmatrix} 2.707 & -0.707 & -2 & 0 \\ -0.707 & 2.707 & 0 & 0 \\ -2 & 0 & 2.707 & 0.707 \\ 0 & 0 & 0.707 & 2.707 \end{bmatrix} \begin{bmatrix} u_{x2} \\ u_{y2} \\ u_{x3} \\ u_{y3} \end{bmatrix}$$

Solving the matrix we get

$$2.707 u_{x2} - 0.707 u_{y2} - 2 u_{x3} = 8 \times 10^{-3}$$

$$-0.707 u_{x2} + 2.707 u_{y2} = 0$$

$$-2 u_{x2} + 2.707 u_{x3} + 0.707 u_{y3} = 8 \times 10^{-3}$$

$$0.707 u_{x2} + 2.707 u_{y2} = 0$$

$$\begin{bmatrix} u_{x2} \\ u_{y2} \\ u_{x3} \\ u_{y3} \end{bmatrix} = \begin{bmatrix} 8.54 \times 10^{-3} \\ 2.23 \times 10^{-3} \\ 6.77 \times 10^{-3} \\ -1.77 \times 10^{-3} \end{bmatrix}$$